

# The ST7 Interferometric Displacement Sensor - a progress report -

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Aspen 2003, Gravitational Wave Advanced  
Detector Workshop, February 2-8, 2003

LIGO-G030211-00-Z



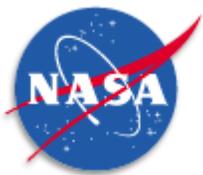
## ST7 Interferometer people

- Bill Folkner, ST7 PI
- Guy Man, ST7 PM
- -----
- Allan Sirota, Interferometer PEM
- Robert Spero
- Jake Chapsky
- Rosemary Diaz
- Steve Azevedo
- A.K., Interferometer CogE



# Acknowledgement

This Research was carried out at the Jet Propulsion Laboratory,  
California Institute of Technology, under contract with the  
National Aeronautics and Space Administration.



# Disturbance Reduction System

NMP

- Technology validation of sensor and thrust-producing technologies to control a space vehicles flight path so the payload responds only to gravitational forces.

*Sensor: Stanford University, Stanford, CA*

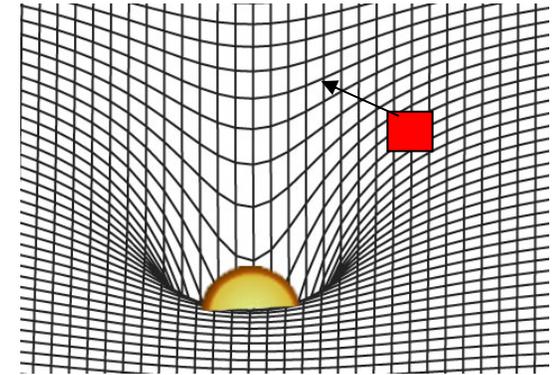
*Thruster: Busek Company Inc. Natick, Mass.*

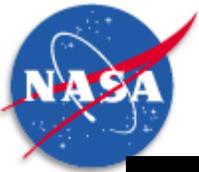
- Launch 2006 as NASA's Space Technology 7 project (ST7)
  - Piggy-backing on ESA's SMART-2 Mission



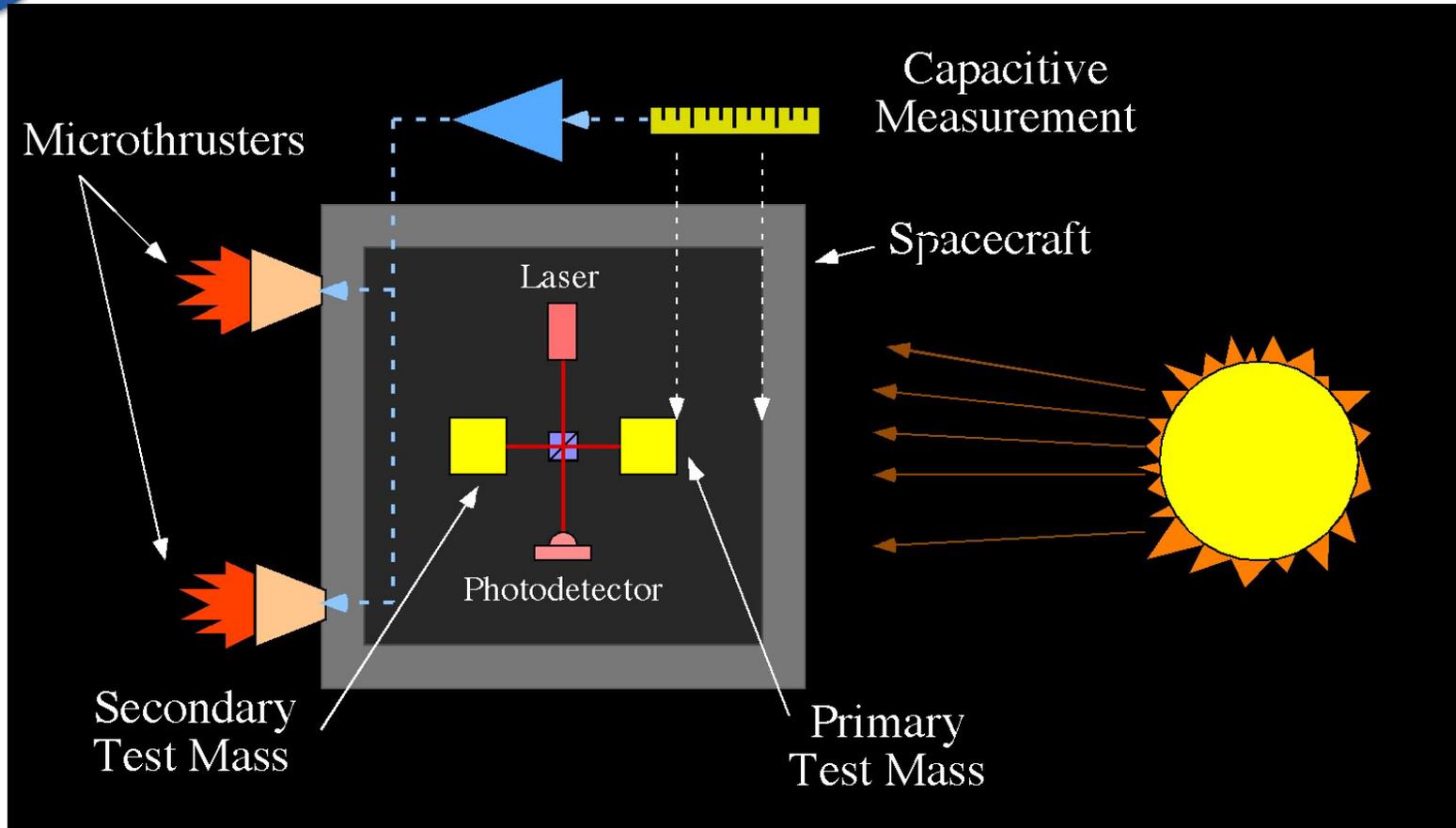
# ST7 Technology Objectives

- Validate that a test mass follows a trajectory determined by gravitational forces only within  $3 \times 10^{-14} \text{ m/s}^2/\sqrt{\text{Hz}}$ 
  - Low acceleration noise is needed for study of general relativity, planetary gravity, gravitational waves
  
- Validate spacecraft position control to an accuracy of  $<10 \text{ nm}/\sqrt{\text{Hz}}$ 
  - Spacecraft position control is required for separated-spacecraft interferometers which do not use internal delay lines

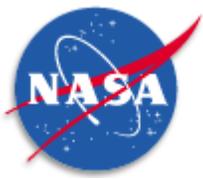




# ST7 Concept



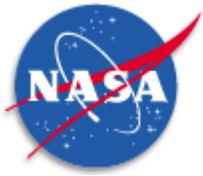
- The DRS instrument package consists of
  - Two gravitational reference sensors
  - Microthrusters for spacecraft position control
  - Interferometer to measure the distance between the two test masses.



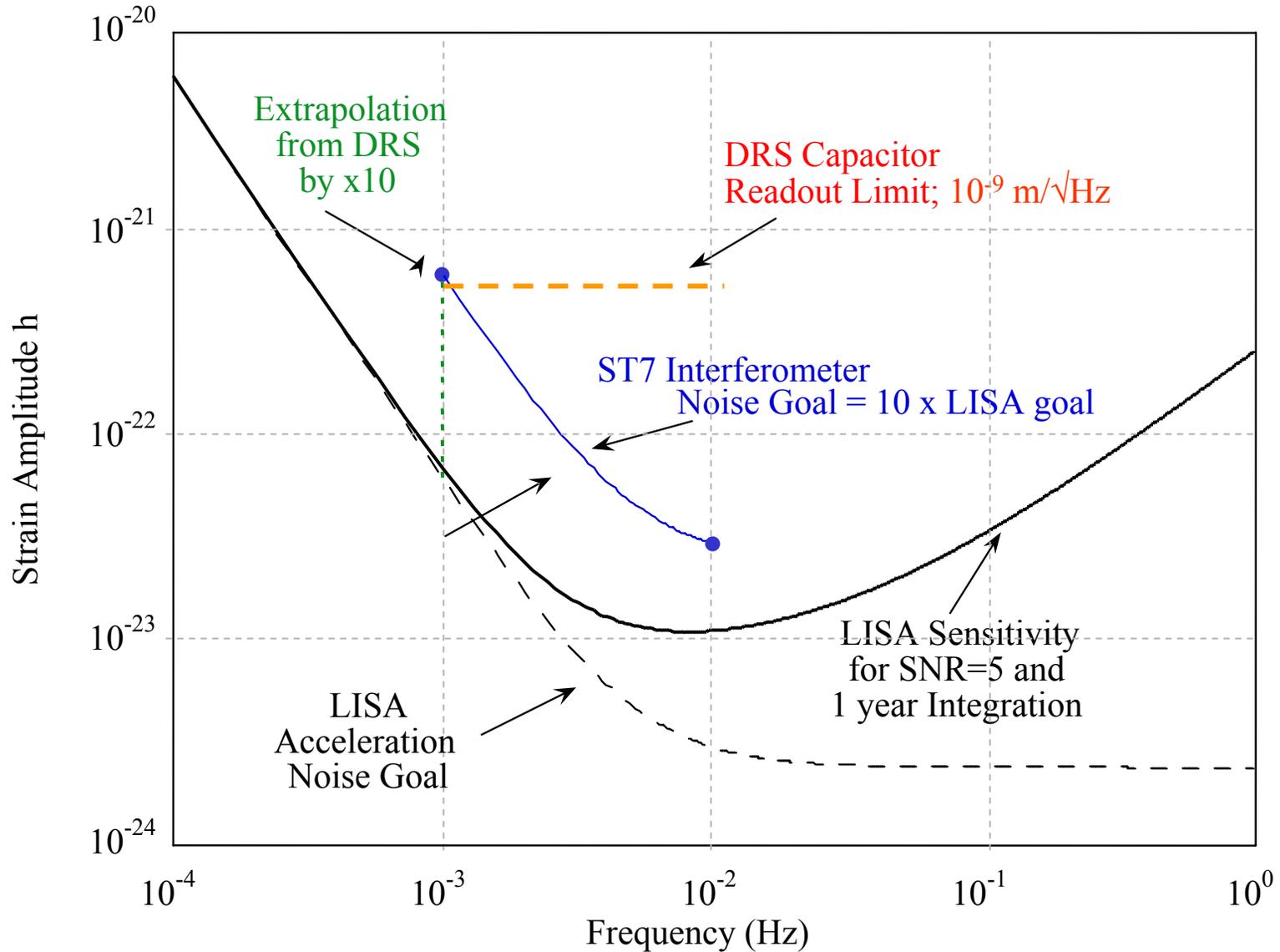
# Required DRS Performance

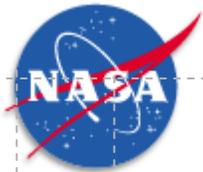
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- LISA requirement @ 1 mHz:  $3 * 10^{-15} \text{ m/s}^2/\sqrt{\text{Hz}}$
- ST7 goal @ 1 mHz:  $3 * 10^{-14} \text{ m/s}^2/\sqrt{\text{Hz}}$
- Proof mass sensor noise:  $1 \text{ nm}/\sqrt{\text{Hz}}$
- Interferometer noise requirement :  $280 \text{ pm}/\sqrt{\text{Hz}}$  @ 1 mHz  
(1/3 of error budget)
  
- LISA requirement @ 10 mHz:  $1.5 * 10^{-14} \text{ m/s}^2/\sqrt{\text{Hz}}$
- ST7 goal @ 10 mHz:  $1.5 * 10^{-13} \text{ m/s}^2/\sqrt{\text{Hz}}$
- Interferometer noise requirement:  $30 \text{ pm}/\sqrt{\text{Hz}}$  @ 10 mHz  
(1/3 of error budget)
  
- $a = (2\pi f)^2 * 5 * 10^{-10} \text{ m}/\sqrt{\text{Hz}} \sim 2 * 10^{-14} \text{ m/s}^2/\sqrt{\text{Hz}}$  @ 1 mHz
- $a = (2\pi f)^2 * 1 * 10^{-9} \text{ m}/\sqrt{\text{Hz}} \sim 4 * 10^{-14} \text{ m/s}^2/\sqrt{\text{Hz}}$  @ 1 mHz
- $4 * 10^{-12} \text{ m/s}^2/\sqrt{\text{Hz}}$  @ 10 mHz)
- $a = (2\pi f)^2 * 5 * 10^{-11} \text{ m}/\sqrt{\text{Hz}} \sim 2 * 10^{-13} \text{ m/s}^2/\sqrt{\text{Hz}}$  @ 10 mHz

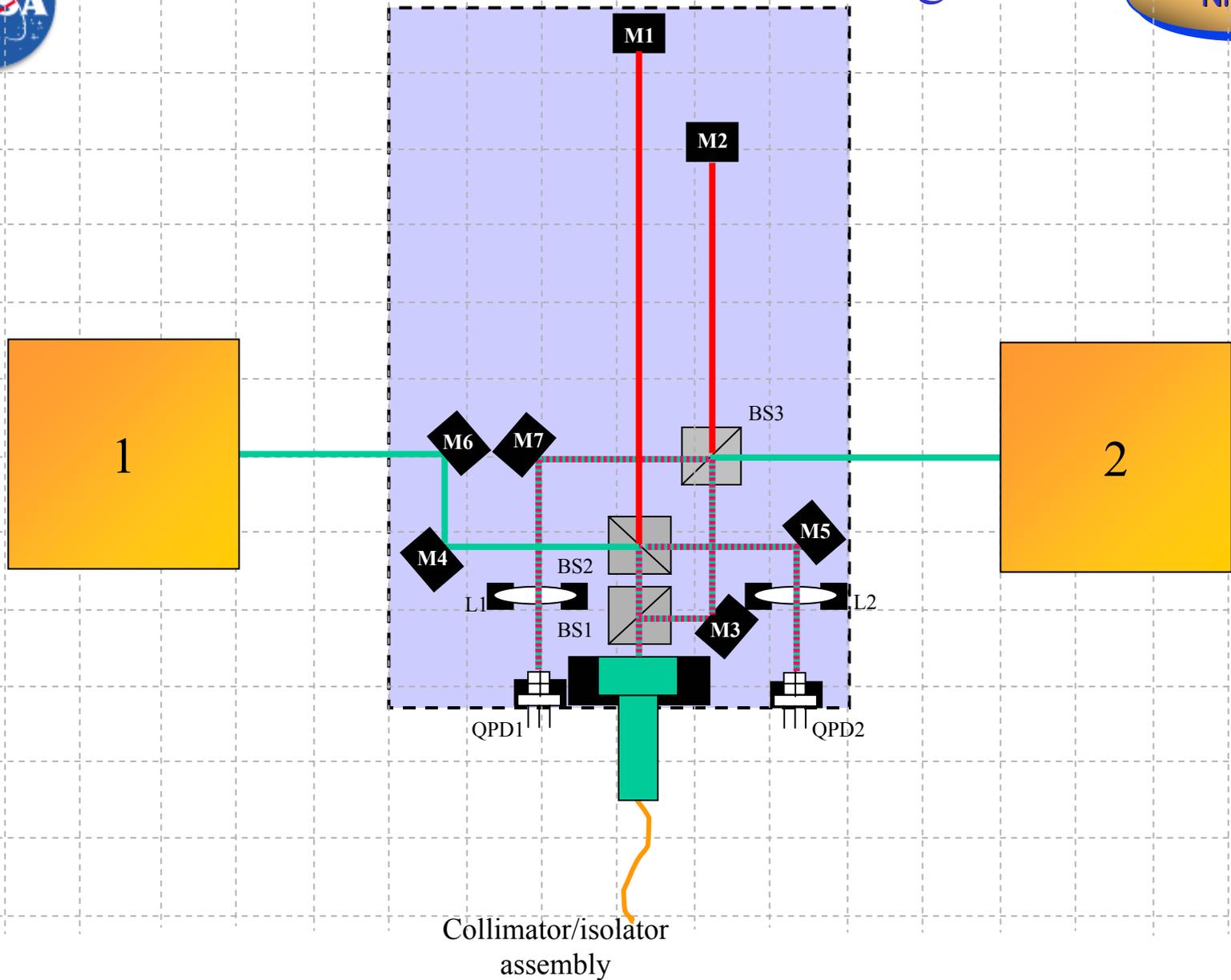


# DRS Performance Limits





# Interferometer Baseline Design

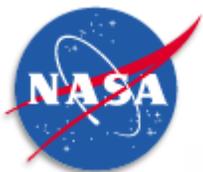




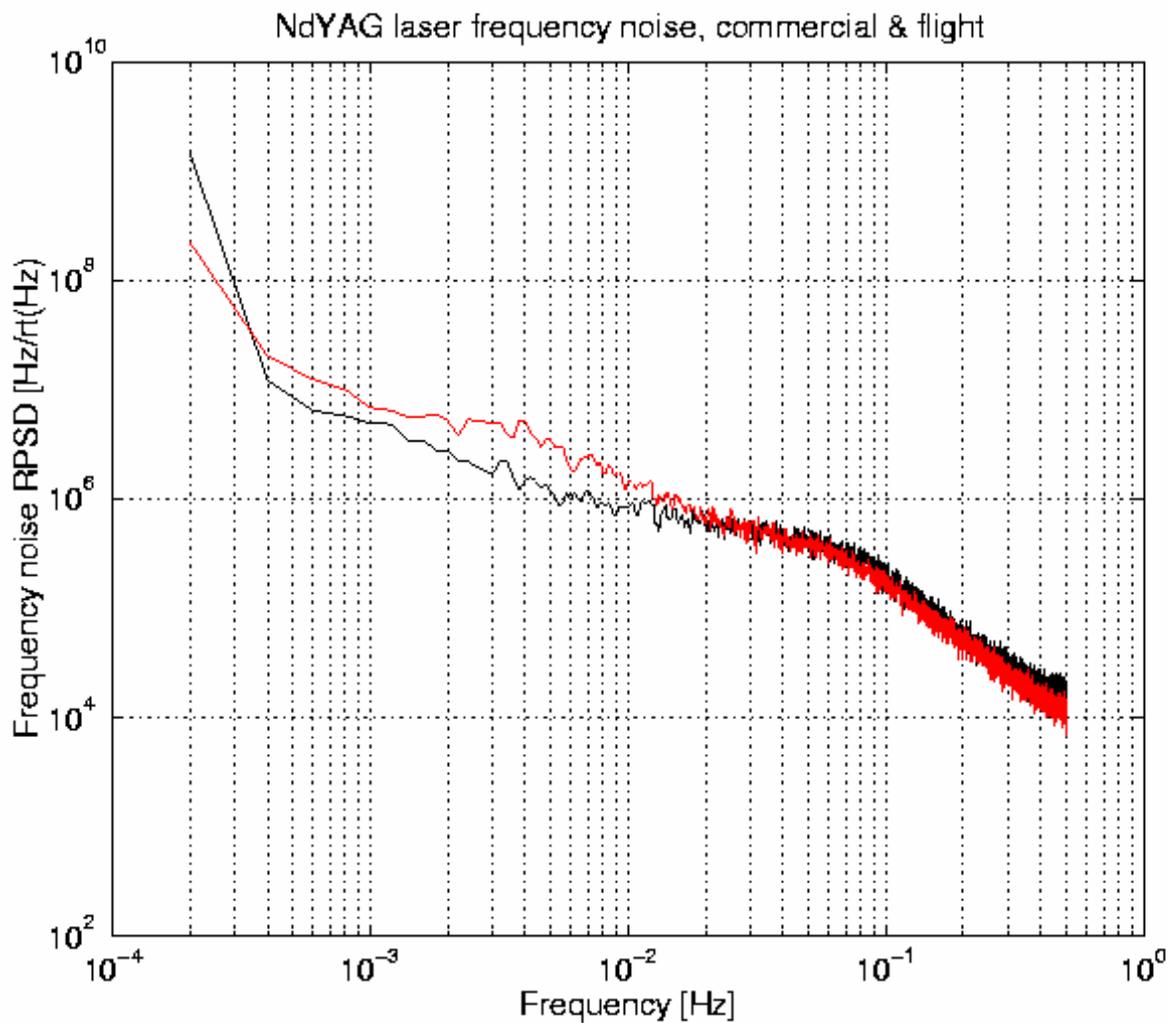
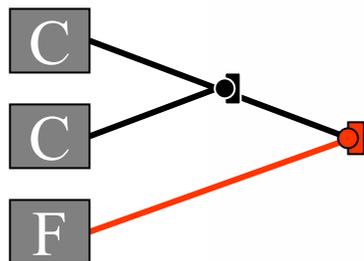
# Laser Frequency Noise

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- Flight laser noise measured:
    - $\sim 1.5 \text{ MHz}/\sqrt{\text{Hz}}$  @ 10 mHz
    - $\sim 7 \text{ MHz}/\sqrt{\text{Hz}}$  @ 1 mHz
  - Interferometer pathlength mismatch:
    - $\Delta L \sim 0.5 \text{ mm}$
  - Laser noise  $\rightarrow$  path noise:
    - @ 10 mHz
      - $\tilde{x}(f) = \Delta L * \tilde{v}(f) / v_0$
      - $\sim 3 \text{ pm}/\sqrt{\text{Hz}}$
    - @ 1 mHz
      - $\sim 12 \text{ pm}/\sqrt{\text{Hz}}$
- Compare to:
- Interferometer noise req'd:
    - $30 \text{ pm}/\sqrt{\text{Hz}}$  @ 10 mHz
  - Interferometer noise req'd:
    - $280 \text{ pm}/\sqrt{\text{Hz}}$  @ 1 mHz

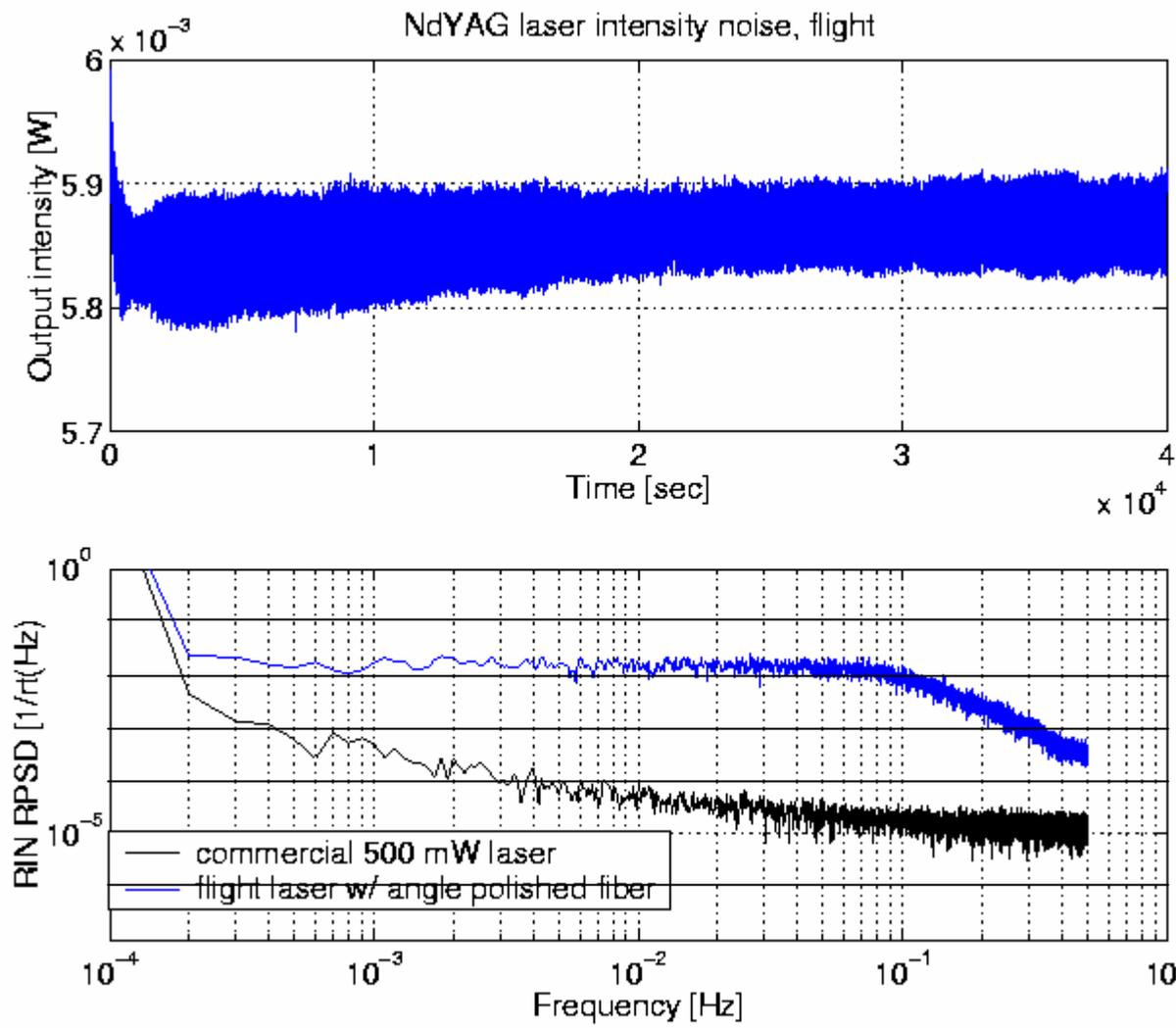


# Laser frequency noise



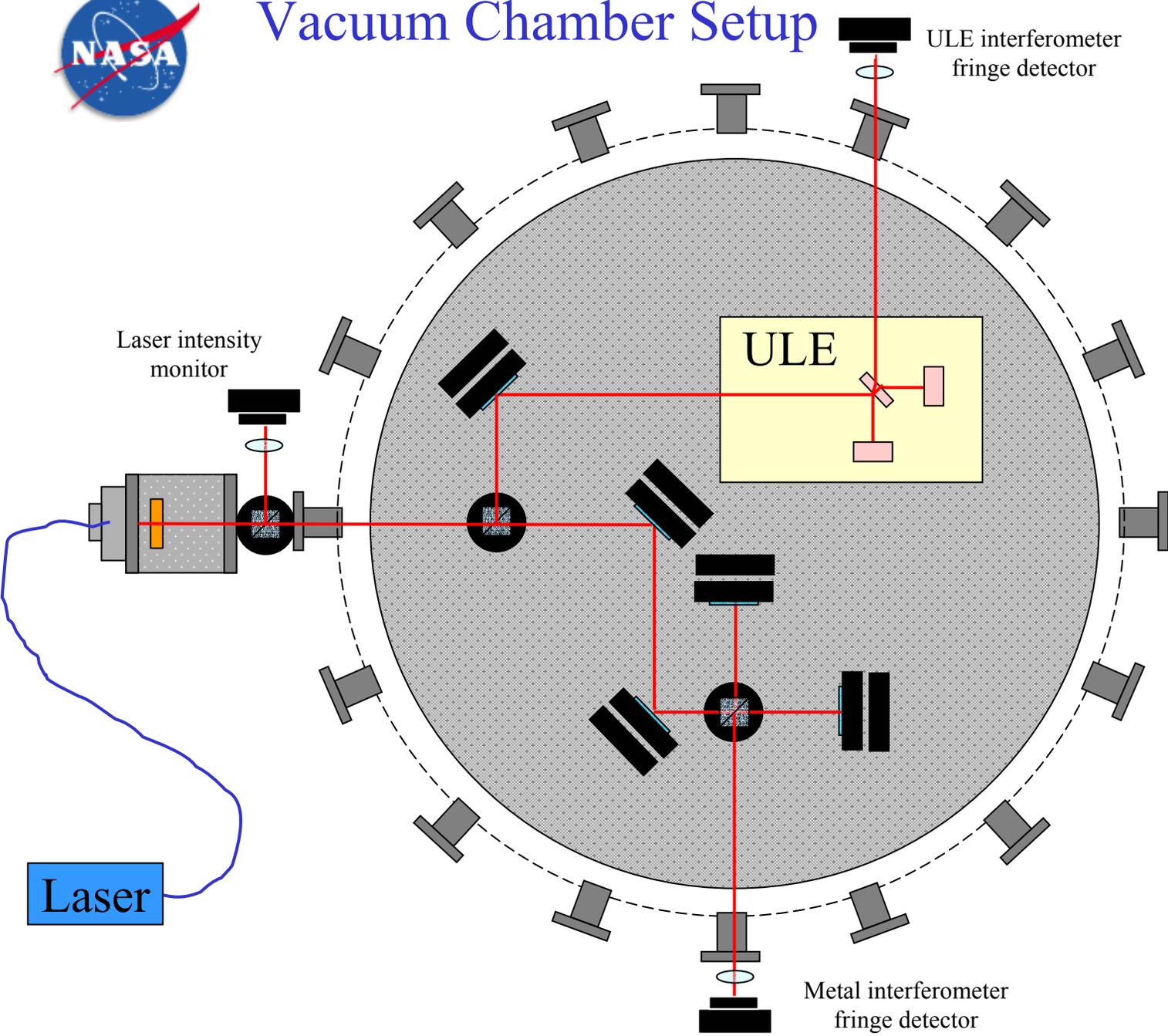


# Laser intensity noise





# Vacuum Chamber Setup



Laser



# Interferometer Results



Metal  $d < 1$  mm; ULE  $d = 4$  mm; t012103d;  $0 < t < 7.75$  hr-NS4; BW = 212 uHz

